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TM(L)-1124 000 00

Satellite Computer Program Description

Milestone 4

Compute Shadow Times

(SHADOW)

TECHNICAL MEMORANDUM

(TM Series)

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Satellite Computer Program Description	SYSTEM
Milestone 4	DEVELOPMENT
Compute Shadow Times (SHADOW)	CORPORATION
by	2500 COLORADO AVE.
C. M. Seacat	SANTA MONICA
22 March 1963	CALIFORNIA
Approved	
J. D. Marioni	

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INTRODUCTION

SHADOW is a 1604 COP-controlled program designed to compute the times at which a satellite enters and leaves the shadow of the earth.

1.0 IDENTIFICATION

Compute shadow times (SHADOW)

2.0 FUNCTION

The inputs to this program are a reset tape containing orbital conditions and a function card specifying a time range and a time increment. The latter is the time between computations and is equivalent to accuracy desired. The output is an on-line printout of shadow periods (see Section 6.0).

3.0 MATHEMATICAL FORMULATION

The following assumptions are made:

- a. The shadow of the earth is cylindrical and non-fuzzy.
- b. The positions of the satellite and the sun are known for a given time, t , in inertial coordinates.

The coordinates of the sun at time, t , are (x,y,z) . The axis of the cylinder formed by the shadow of the earth passes through the point $(-x,-y,-z)$.

The coordinates of the satellite at time, t , are (ξ, η, ρ) .

The equation $\rho^2 = \xi^2 + \eta^2 + \zeta^2$ defines an earth-centered sphere of radius, ρ , with the satellite on the surface.

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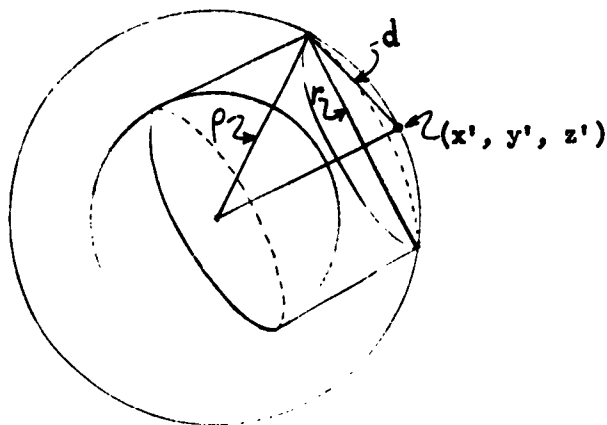
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The axis of the cylindrical shadow pierces this sphere at x' , y' , z' ,
where

$$\begin{aligned}x' &= -P/R \ x \\y' &= -P/R \ y \\z' &= -P/R \ z \\R &= \sqrt{x^2 + y^2 + z^2}\end{aligned}$$

Graphically, the situation can be represented as follows:



The distance, d , from (x', y', z') to the circular intersection of the cyclinder and the sphere is given by:

$$d^2 = 2\rho(\rho - \sqrt{\rho^2 - r^2}) \quad \text{where } r = \text{earth radius}$$

The distance, D , from (x', y', z') to the satellite is given by

$$D^2 = (x' - \xi)^2 + (y' - \eta)^2 + (z' - \zeta)^2$$

If $D^2 > d^2$, the satellite is in sunlight.

If $D^2 \leq d^2$, the satellite is in shadow.

Please note that the analysis above does not assume a circular or near-circular orbit. A new ρ is computed for each t .

4.0 CALLING SEQUENCE

This program is normally called by a function card with the format:

* SHADOW	A	B	C	D	E	F.F	G	H	I	J	K.K	L.L
----------	---	---	---	---	---	-----	---	---	---	---	-----	-----

A	Vehicle number	
B	Month	
C	Day	
D	Hour	Start Time
E	Minute	
F.F	Second	
G	Month	
H	Day	
I	Hour	Stop Time
J	Minute	
K.K	Second	
L.L	Computation interval in seconds	

An option exists to specify start and stop times by rev number. The function card has a format:

* SHADOW A B. C. D.D

A Vehicle Number
B Starting Rev (decimal point mandatory)
C Ending Rev (decimal point mandatory)
D.D Computation interval in seconds

A fractional rev number may be input without harming the program, but the fractional part will be ignored. For consistency between the function card and printed output, it is recommended that zero, or blank, follow the decimal point in the rev number.

Computation is carried out up to, but not including, the terminal rev.

The last field on the card, computation interval, controls the accuracy of the printout and the time consumed by the program. For example, if the computation interval were 60, the printout would be accurate to within one minute. If the computation interval were 1, the printout accuracy would be one second and the program would take roughly sixty times longer. Crude timing estimates indicate that for a computation interval of sixty seconds, the program requires between one and two seconds per rev.

5.0 DETAILED PROCESSES

The program computes the inertial coordinates of the satellite at time, t , by successive branches to the subroutines TTE and PTR. The coordinates of the sun, obtained by table look-up, are taken from the American Ephemeris and Nautical Almanac and contain one value of x , y , and z for each day of the year in the following format.

Location	Contents
L	x
L+1	y December 31, preceding year
L+2	z

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Location	Contents
L+3	x
L+4	y January 1, current year
L+5	z
.	.
.	.
.	.
L+1095	x
L+1096	y December 31, current year
L+1097	z
L+1098	x
L+1099	y January 1, next year
L+1100	z

To get the sun coordinates at a given time, t, the program performs a linear interpolation between successive triads of tabular (x, y, z).

The entries in the table can be transcribed directly from the almanac. Assuming the first address of the table to be tagged SUNCOORD, the first few instruction cards for the 1963 version of the table would be:

Columns	1 123456789012	2 012345678	4 1 (Comments field)
	SUNCOORD DEC	+.1509848	x, Dec 31
	DEC	-.8914589	y, Dec 31
	DEC	-.3865841	z, Dec 31
	DEC	+.1682445	x, Jan 1
	DEC	-.8888335	y, Jan 1
	DEC	-.3954462	z, Jan 1
		.	
		.	
		.	
		(etc.)	

6.0 OUTPUTS

Every time the satellite leaves the shadow of the earth, the program prints on line the date, time, rev number, latitude (geodetic), and longitude at which it entered and left. When the stop time is reached, the program prints out the percentage of time in shadow (i.e., the sum of all the shaded periods divided by the time range on the function card). A sample printout is shown in Appendix A.

7.0 INTERFACES AND RESTRICTIONS

The program uses the following subroutines.

Code name

RESET

PTR

LTT

TTE

SETUP

SUBERR

RADDEG

OUTPUT

FLOAT

FIX

RALONG

UTIMAC

MACGUT

SQRT

ATAN1

TAN

Once a year the table of sun coordinates (see Section 5.0) must be updated, and the program reassembled.

Reset tapes containing orbital conditions must be mounted on tape drives 2, 9, and 10.

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To start from scratch, the following function sequence is recommended:

*IRT ...

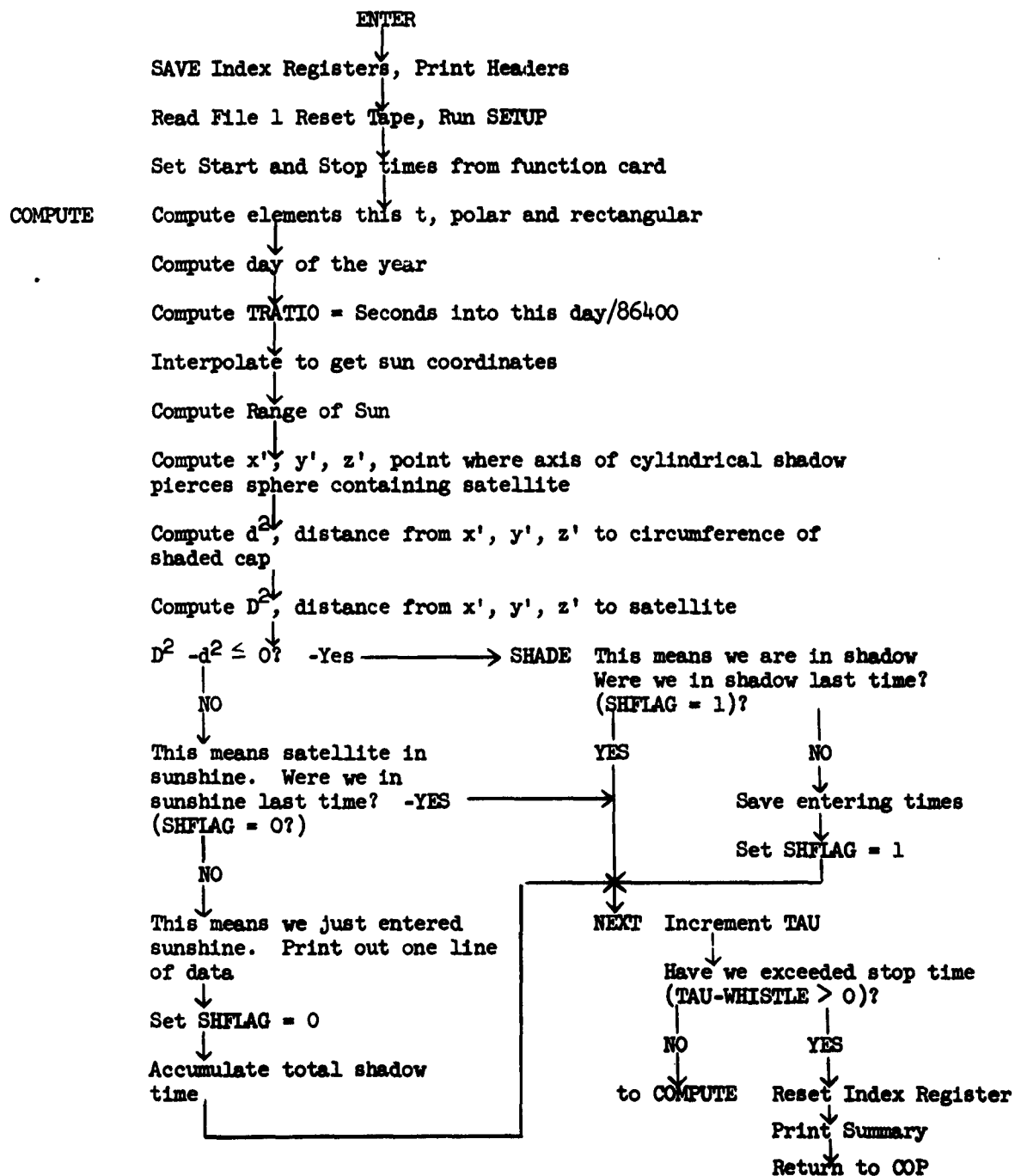
*WNRT ...

*INJFUN...

(Each card completed appropriately)

*SHADOW ...

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APPENDIX A
SAMPLE PRINTOUT

VEH 1136 RUN NO. SAMPLE99
SHADOW INTERVALS TABULATED BELOW

ENTER				LEAVE		ENTER				LEAVE	
MO	DAY	HR	MIN	SEC	SYSTEM	REV	HR	MIN	SEC	SYSTEM	REV
11	1	12	25	0	44700.	9.8	13	0	0	46800.	10.2
11	1	13	55	0	50100.	10.8	14	30	0	52200.	11.3
(etc)											
11	2	10	35	0	38100.	24.8	11	10	0	40200.	25.2
TOTAL SHADOW TIME 35400. SECONDS FOR 41. PERCENT											
						DURATION		GEOD LAT		GEOD LAT	
						2100.		53 15.9S		88 59.6N	
						2100.		47 49.4S		85 35.9N	
								LONG		LONG	
								179 14.2E		170 27.8E	
								156 40.5E		32 5.9W	
								154 10.4W		89 30.5N	
								2100.		162 56.9W	

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System Development Corporation,
Santa Monica, California
SATELLITE CONTROL PROGRAM DESCRIPTION
MILESTONE 4 COMPUTE SHADOW TIMES (SHADOW)
Scientific rept.; TM(L)-1124/000/00, by
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(Contract AF 19(628)-1648, Space
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Unclassified report

DESCRIPTORS: Satellite Networks.
Programming (Computers).

States that SHADOW is a 1604 COP-
controlled program designed to compute

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the times at which a satellite enters
and leaves the shadow of the earth.
Also states that the inputs to this
program are a reset tape containing
orbital conditions and a function card
specifying a time range and a time
increment.

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